

Federal Aviation Administration

Rotorcraft Damage Tolerance R&D

FAA Rotorcraft
Structural Integrity and
Safety

Presented to: Rotorcraft Community

By: Dy Le – FAA Rotorcraft Program manager

Date: March 21, 2006



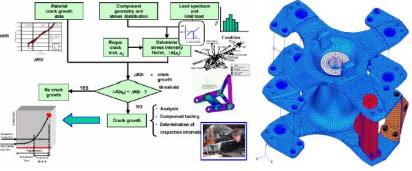
Agenda - Presentation Outline

- → FAA Rotorcraft Structures Research Overview
- →RCDT R&D History and Status
 - → Joint RCDT Efforts
 - → RCDT Accomplishment
 - → RCDT Technology Readiness Level Status
- →Overview of the Development of RCDT R&D Strategic Plan and Roadmap
- → FAA RCDT R&D Roadmap
- → FAA RCDT & HUMS BAAs Status Q&A

Rotorcraft Structural Integrity and Safety











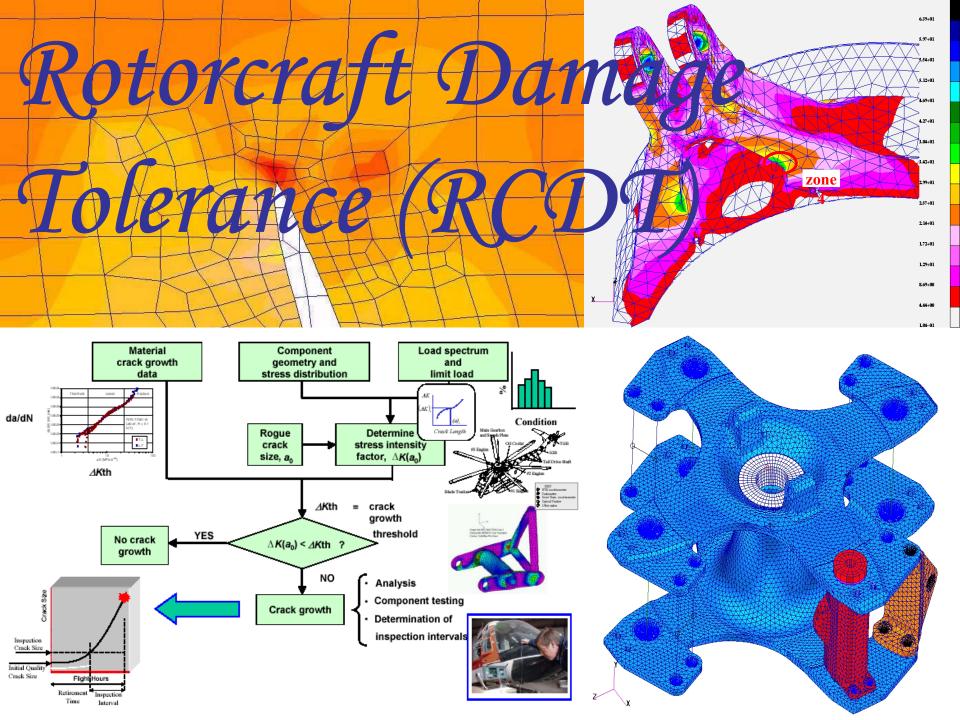


Partnerships/Collaborations in Rotorcraft Structures Research





A joint rotorcraft program with industry, academia, DoD, and other government research agencies to develop and validate mutually agreed-upon technologies that will address rotorcraft structures design and certification issues.





RCDT R&D Roadmap

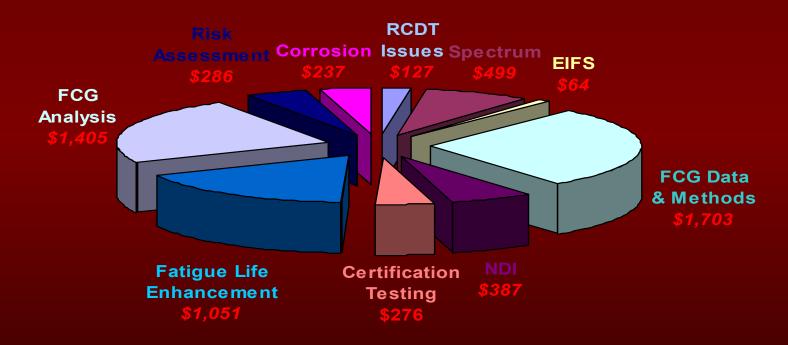




RI N

						UIII MARTO
		2000	2001	2002	2003 —	→ 2005
	FA	A Rulemaking	FAA Metallic	Structures Rulemaki	ng & AC writing	NPRM
				A A	1 1	
		1 RC	CDT Specific Issues	Study	Case Studi	es
	2	Spectrum Dev	velopment	Usage Monitorin	Usage N	Monitoring Dev.
>					/ Y	
Ţ.			EIFS I	<u>Database</u>	Databa	se Dev.
Ţ.						
decreasing priority	4 (Coupon Test Method	K-th, Long & Short	Crack, HCF, & Peened	Coupon Database	Database Dev.
ng						l N
asi	5	NDI Survey / Po	echnology Developi	ment		
cre						
de			6 Certification Te	st Guidelines Analy	vsis Alternatives to C	Cert. Tests Dev.
			- I:C D 1			
			7 Life Enhance	ement Methods	Life Enhan	cement R&D
	0	CCAD /III C	A 1 C 1 HOE/LO	F	D 1 1 C	WED DOD
	8	CGA Dev/Val; Stress	Anal Guides; HCF/LC	F interaction; K-sol'ns	Residual Stre	ss/WFD R&D
		Support advisory materia	1 9 Risk Assessm		D., -1, -1, :1: -4	i- DT D & D
		_	9 KISK ASSESSM	ent Case Studies	Probabilist	ic DT R&D
1	, <mark>L</mark>	Compliance issues				
•		Technology enhancemen	ts 10	Corrosion	Control R&D	
		_ 	 			
		Industry D	roduct Dovolonment	t and Field Support T	Cashnalagy Implam	entation
		Illuusu y F	Toduct Developmen	t and Field Support 1	echnology impleme	entation
		Composite S	tructures DT R&D,	Rulemaking and Ad	visory Circular (AC) Material
		I .	1	I and the second	1	

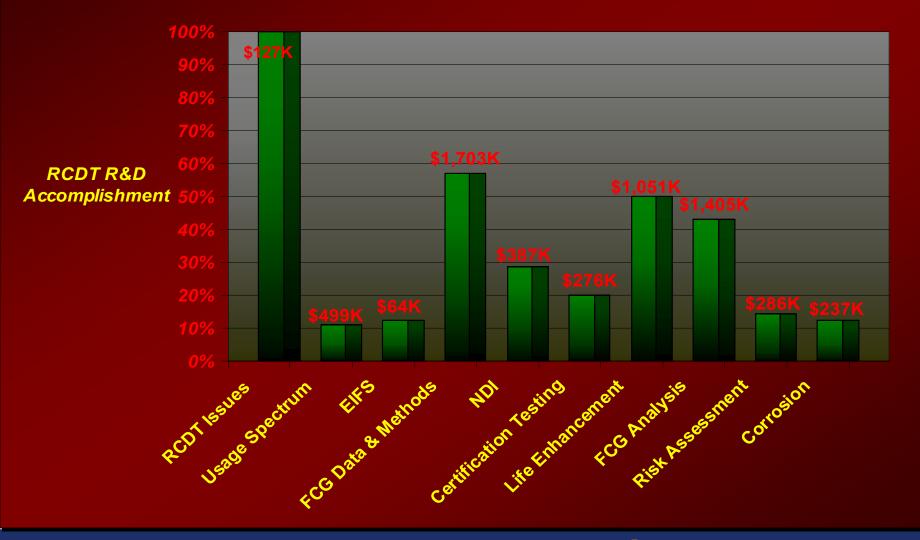
FAA RCDT Phase I R&D Funding (\$K)



FY00 - FY05 (\$6M)

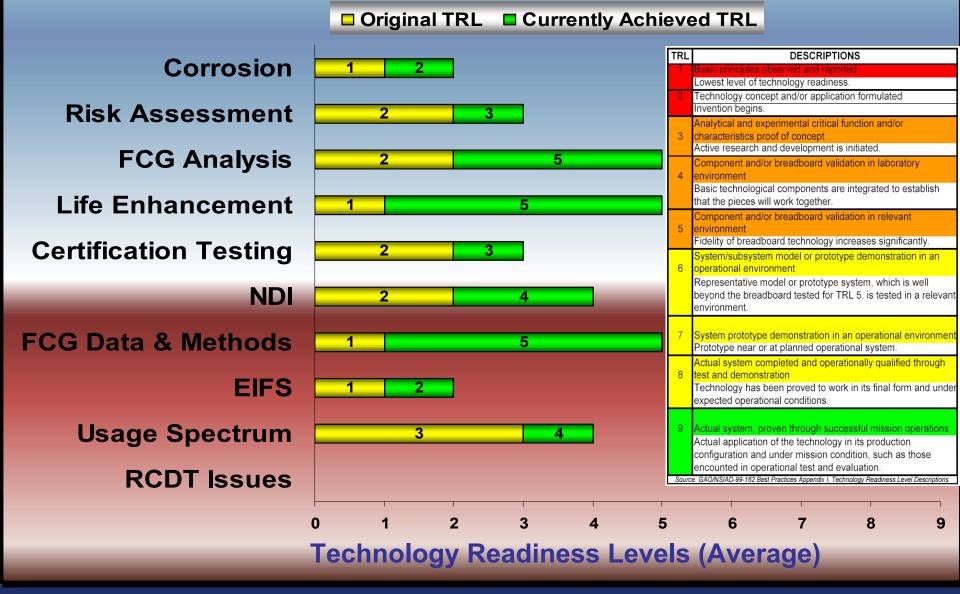


FAA RCDT Research Status and Accomplishment FY00 - FY05 (\$6M)





RCDT TRL Status (FY00 - FY05)



RCDT R&D STRATEGIC PLAN DEVELOPMENT INITIATIVES

- → Rotorcraft Community RCDT and Fracture Mechanics R&D Perspectives
 - → OEM's, Academia, Government
- → FAA's RCDT R&D Perspectives
 - → FAA-Rotorcraft Community combined assessment of current state of RCDT technologies
 - → FAA's process to identify and prioritize RCDT R&D needs

FAA RCDT R&D Plan Development

- → RCDT technology assessment and gap analysis
 - → RCDT technology matrices
 - → TRL criteria
- → Identified R&D needs
 - → FAA regulation support criteria
- → Development of RCDT R&D Roadmap
 - > FAA R&D prioritization criteria
- → Development of RCDT R&D strategic plan
 - → Output requirements & performance metrics
 - → exit criteria/strategy
 - → ROM

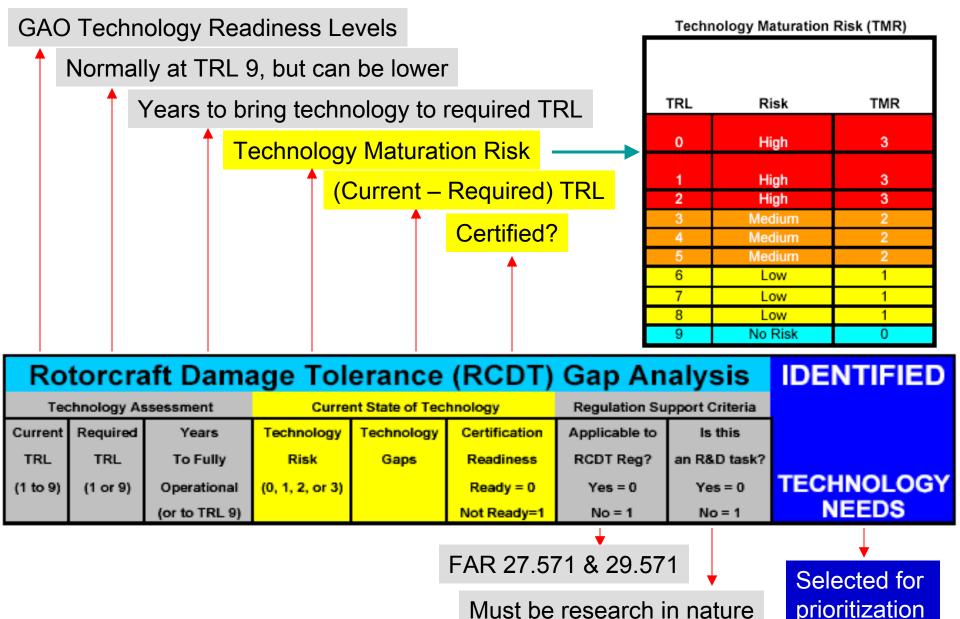


RCDT Technology Matrices

- → Usage and Load Spectrum
- → Initial Crack State
- → Material Fatigue Crack Growth (FCG) Data
- → Fracture Mechanics (FM) and FCG Analyses
- → Risk Assessment Probabilistics
- → Non-Destructive Inspections and Evaluations
- → RCDT Demonstration and Compliance
- → RCDT Documentation
- → Regulatory Implementation and Technology Transfer

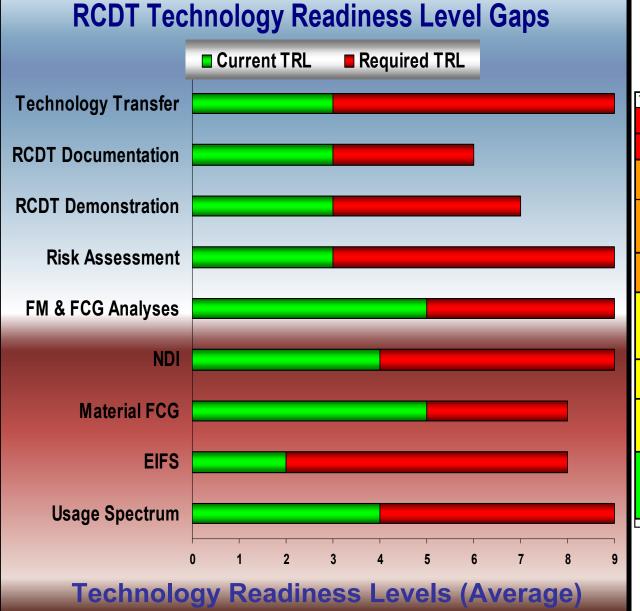
		TION							
	Ro	torcra	aft Dama	age Tol	erance	(RCDT)	Gap An	alysis	IDENTIFIED
	Tec	hnology As	sessment	Curre	nt State of Teci	hnology	Regulation Su	pport Criteria	
	Current	Required	Years	Technology	Technology	Certification	Applicable to	Is this	
	TRL	TRL	To Fully	Risk	Gaps	Readiness	RCDT Reg?	an R&D task?	
	(1 to 9)	(1 or 9)	Operational	(0, 1, 2, or 3)		Ready = 0	Yes = 0	Yes = 0	TECHNOLOGY
	, ,	, ,	(or to TRL 9)	, , , , , , , ,	Not Ready		No = 1	No = 1	NEEDS
OVERALL RCDT TECHNOLOGY STATUS	4	7	3	2	3	0	98%	96%	
Component Stress Distribution	4	7	3	2	3	0%	100%	100%	
Finite Element Methods and Codes	4	7	3	2.0	3	0%	100%	100%	
Enhancement of FEA complex geometry, non-linear stress field	5	7	3	2	2	1	0	0	FEA Enhancement
and fracture mechanics codes	- J								
(integration)	3	7	3	2	4	1	0	0	FEA & FM Interface
Usage and Load Spectrum	4	9	3	2	4	0%	100%	100%	
Usage Data Collection		:	8	2	8	0%	100%	100%	
Establishment of Typical Spectrum Using HUMS	4	9	3	2	5	1	0	0	Establishment of Typical Spectrum Using HUMS
Usage monitoring using HUMS	7	9	3	1	2	1	0	0	Usage Monitoring
Sensitivity of FCG to usage/load spectra variations	5	7	2	2	2	1	0	0	FCG & Load Sensitivity
Load Spectrum	4	9		2	-	0%	100%	100%	
Heavy lift components strain survey	4	9	5	2	5	1	0	0	Strain Survey
Direct Load Measuring Using HUMS	4	9	5	2	5	1	0	0	Using HUMS
RHL RC Fatigue Life	4		2	2	•	0%	100%	100%	Deposited House Life
Initial Crack Size State	3	9	3	2 2	5 3	0	100%	0 89%	Repeated Heavy Lift
Determination of Equivalent Initial Flaw	3	:	3	9		0%	100%	89%	
Establishment of RC Damage Database	2	6	5	3	4	1	0	0	Damage Database
Update FCG models to make EIFS be independent load levels	4	6	2	2	2	1	0	0	FCG Models
Assessment of Relliability and Maintainability (R&M) FRACAS and RCDT damage databases	4	6	2	2	2	1	0	1	No
Determination of Normal Quality and Rogue EIFS Distribution	2	6	5	3	4	1	0	0	Quality and Rogue EIFS Distribution
Characterization of small initial damage states	2	8	3	3	6	1	0	0	Characterization of small initial damage states
development and application of EIFS to RCDT design and management for manufacturing and in-service damage threats	2	6	2	3	4	1	0	0	EIFS Development Guidelines

RCDT ASSESSMENT AND GAP ANALYSIS





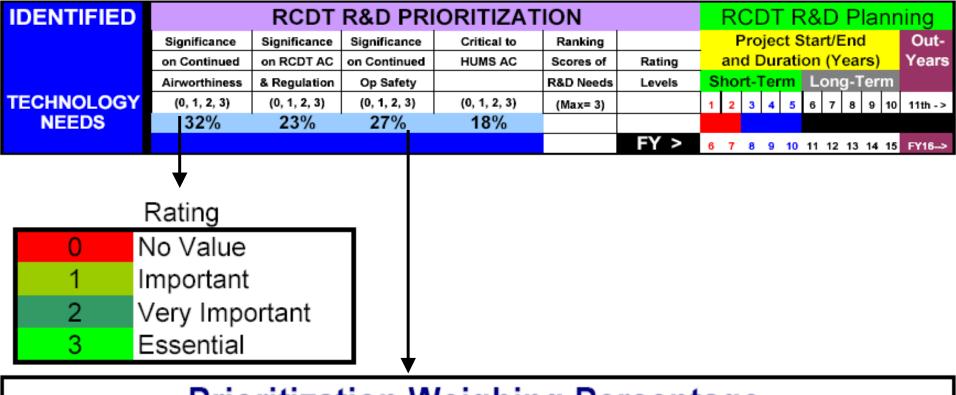
		TION							
	Ro	torcra	aft Dama	age Tol	erance	(RCDT)	Gap An	alysis	IDENTIFIED
	Tec	hnology As	sessment	Curre	nt State of Teci	hnology	Regulation Su	ipport Criteria	
	Current	Required	Years	Technology	Technology	Certification	Applicable to	Is this	
	TRL	TRL	To Fully	Risk	Gaps	Readiness	RCDT Reg?	an R&D task?	
	(1 to 9)	(1 or 9)	Operational	(0, 1, 2, or 3)		Ready = 0	Yes = 0	Yes = 0	TECHNOLOGY
	, ,	, ,	(or to TRL 9)			Not Ready=1	No = 1	No = 1	NEEDS
OVERALL RCDT TECHNOLOGY STATUS	4	7	3	2	3	0	98%	96%	
Component Stress Distribution	4	7	3	2	3	ā	100%	100%	
Finite Element Methods and Codes	4	7	3	2.0	3	0	100%	100%	
Enhancement of FEA complex geometry, non-linear stress field	5	7	3	2	2	1	0	0	FEA Enhancement
and fracture mechanics codes (integration)	3	7	3	2	4	1	0	0	FEA & FM Interface
Usage and Load Spectrum	4	9	3	2	4	0	100%	100%	T EX G T M INICIACC
Usage Data Collection		•	3	2	8	0	100%	100%	
Establishment of Typical Spectrum Using HUMS	4	9	3	2	5	1	0	0	Establishment of Typical Spectrum Using HUMS
Usage monitoring using HUMS	7	9	3	1	2	1	0	0	Usage Monitoring
Sensitivity of FCG to usage/load spectra variations	5	7	2	2	2	1	0	0	FCG & Load Sensitivity
Load Spectrum	4	9	-	2		0	100%	100%	
Heavy lift components strain survey	4	9	5	2	5	1	0	0	Strain Survey
Direct Load Measuring Using HUMS	4	9	5	2	5	1	0	0	Using HUMS
RC Fatigue Life	4	9	2	2	5	1	0	100% 0	Repeated Heavy Lift
Initial Crack Size State	3	6	3	2	3	0	100%	89%	Repeated Heavy Litt
Determination of Equivalent Initial Flaw	3	6	3	2	3	0	100%	89%	
Establishment of RC Damage Database	2	6	5	3	4	1	0	0	Damage Database
Update FCG models to make EIFS be independent load levels	4	6	2	2	2	1	0	0	FCG Models
Assessment of Relliability and Maintainability (R&M) FRACAS and RCDT damage databases	4	6	2	2	2	1	0	1	No
Determination of Normal Quality and Rogue EIFS Distribution	2	6	5	3	4	1	0	0	Quality and Rogue EIFS Distribution
Characterization of small initial damage states	2	8	3	3	6	1	0	0	Characterization of small initial damage states
development and application of EIFS to RCDT design and management for manufacturing and in-service damage threats	2	6	2	3	4	1	0	0	EIFS Development Guidelines



TRL	DESCRIPTIONS
1	Basic principles observed and reported
	Lowest level of technology readiness.
2	Technology concept and/or application formulated
	Invention begins.
_	Analytical and experimental critical function and/or
3	characteristics proof of concept Active research and development is initiated.
	Component and/or breadboard validation in laboratory
4	environment
7	Basic technological components are integrated to establish
	that the pieces will work together.
	Component and/or breadboard validation in relevant
5	environment
	Fidelity of breadboard technology increases significantly.
	System/subsystem model or prototype demonstration in an
6	operational environment
	Representative model or prototype system, which is well
	beyond the breadboard tested for TRL 5, is tested in a relevant environment.
	environment.
7	System prototype demonstration in an operational environment
'	Prototype near or at planned operational system.
	Actual system completed and operationally qualified through
8	test and demonstration
	Technology has been proved to work in its final form and under
	expected operational conditions.
9	Actual system, proven through successful mission operations
	Actual application of the technology in its production
	configuration and under mission condition, such as those
Sour	encounted in operational test and evaluation. e. GAO/NSIAD-99-162 Best Practices Appendix I, Technology Readiness Level Descriptions
Soulo	a. GAO/NGIAD-99-102 Dest l'Hactices Appendix I, Technology ReadinesS Level Descriptions

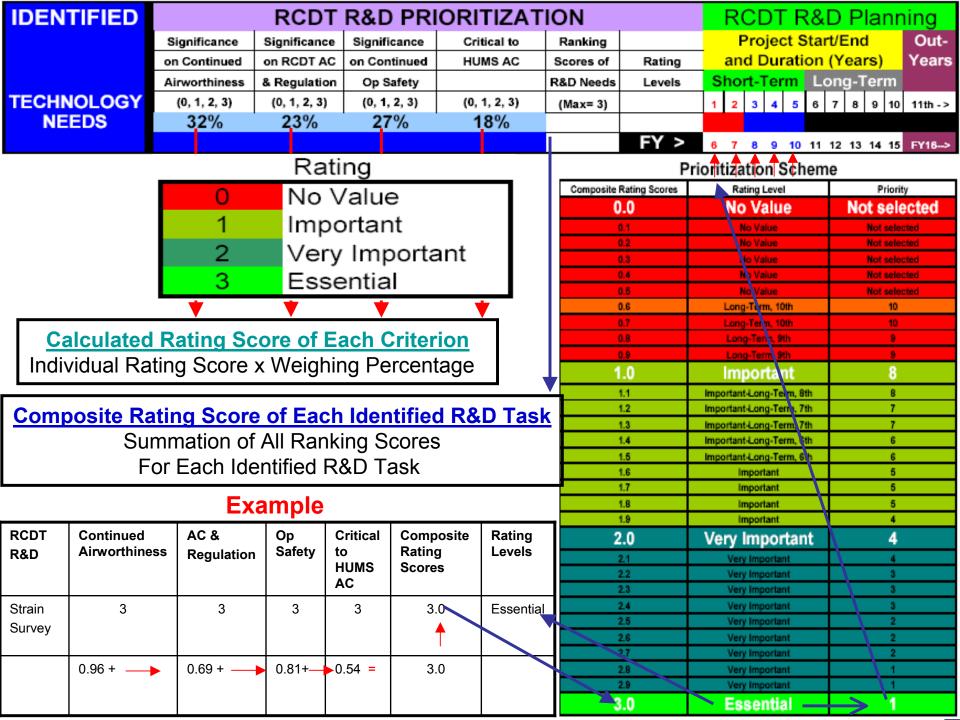


FAA PRIORITIZTION PROCESS



Prioritization Weighing Percentage									
Significance on Continued Airworthiness	Significance on RCDT AC and Regulation	Significance on Continued Operational Safety	Critical to HUMS AC						
32%	23%	27%	18%						

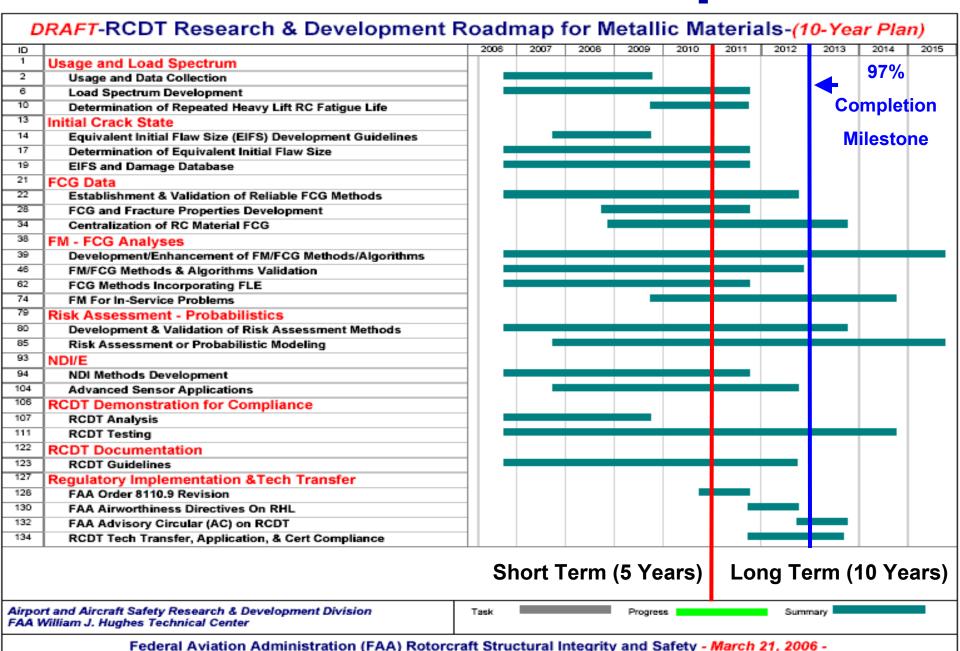


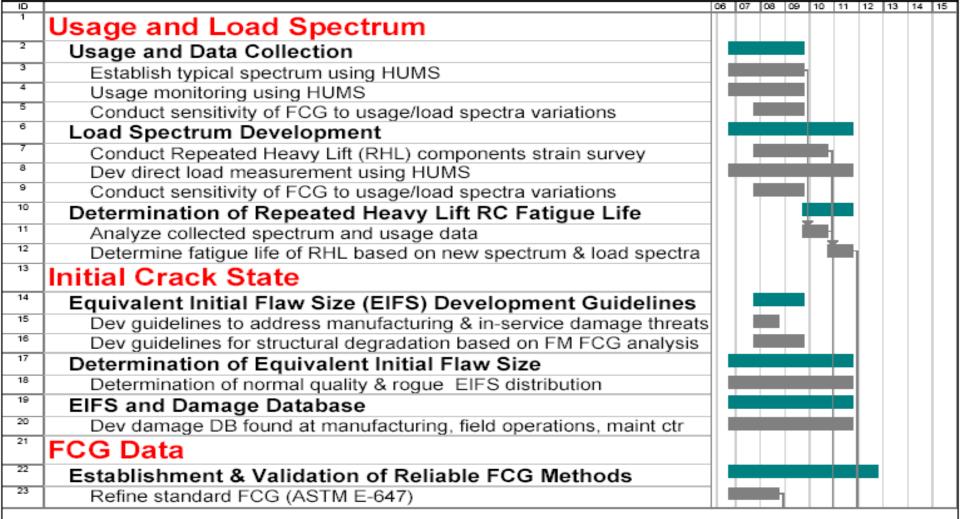


IDENTIFIED	RCDT R&D PRIORITIZATION								RCDT R&D Planning									
	Significance	Significance	Significance	Critical to	Ranking		Project Start/En			nd		Out-						
	on Continued	on RCDT AC	on Continued	HUMS AC	Scores of	Rating		and	Dι	ırat	ion	(Ye	ars)		Years			
	Airworthiness	& Regulation	Op Safety		R&D Needs	Levels	SI	nort	-Te	rm	L	ong	-Ter	m				
TECHNOLOGY	(0, 1, 2, 3)	(0, 1, 2, 3)	(0, 1, 2, 3)	(0, 1, 2, 3)	(Max= 3)		1	2	3 4	6	6	7	8 9	10	11th ->			
NEEDS	32%	23%	27%	18%	, ,													
	0-70		_ , ,	1070		FY >	6	7	R S	10	11	12	13 14	15	FY16->			
	1	1	0	0	1													
	1.0	1.0	0.0	0.0	0.5													
									Т	Т	П	П						
FEA Enhancement	1	1	0	0	0.5	No Value	Н	\dashv	+	╀	╀	\vdash	+	Н				
FEA & FM Interface	1	1	o	0	0.5	No Value												
. 21 0 1 11 11 11 11 12 12	3	2	3	2	3	140 Value												
	8	3	8	8	3													
Establishment of Typical								T	Т	Г								
Spectrum Using HUMS	3	3	3	3	3.0	Essential	3	2	1	+	\vdash	\vdash	+	Н				
Usage Monitoring	3	3	3	3	3.0	Essential	3	2	4	+	\vdash	\vdash	+	Н				
FCG & Load Sensitivity	2	2	3	2	2.3	Very Important	П	2	ıL	ı				П				
TOO a Load Octionary	3	3	3	8	8	very important			1	•								
Strain Survey	3	3	3	3	3.0	Essential	6	4	3 2	1								
Using HUMS	3	3	3	3	3.0	Essential	5	4	3 2	1	\perp		\perp	П				
Dill Feliana Life	2	1	3	1	2	lana antant												
RHL Fatigue Life	2 2.4	1.8	3 2.4	1.1	1.9 2.3	Important			2 1									
	2.4	1.8	2.4	1.1	2.3	_			+	٠	_							
Damage Database	3	3	3	3	3.0	Essential	5	4	3 2	1			\top	П				
Make EIFS independent	1	1	1	1	1.0	Long-Term, 8th	П	Т	Т	Т	П		2 1	П				
make Eiro independent		'	1	'	1.0	Long-Term, och	Н	\dashv	+	┿	╈	\vdash	2 1	Н				
		_					П		Т	ı	П			П				
No Overline and Beauty EIFS	0	0	0	0			Н	+	+	+	\vdash	\vdash	+	Н				
Quality and Rogue EIFS Distribution	3	2	3	1	2.4	Very Important		5	4 3	2	1							
Characterization of small	_					, , , , , , , , , , , , , , , , , , , ,	П		\top	T	Ť			П				
initial damage states	3	2	3	1	2.4	Very Important	Ш	3	2 1	4	\perp			Ш				
EIFS Development																		
Guidelines	3	2	3	1	2.4	Very Important		2	1	L				Ш				
EIFS as Function of																		
Roughness, Flaw Type & Size, Component Geometry	3	2	3	1	2.4	Very Important		2										
Size, Component Geometry	3	2	3	1	2.4	very important		2	1									

FAA RCDT R&D ROADMAP

RCDT R&D Roadmap





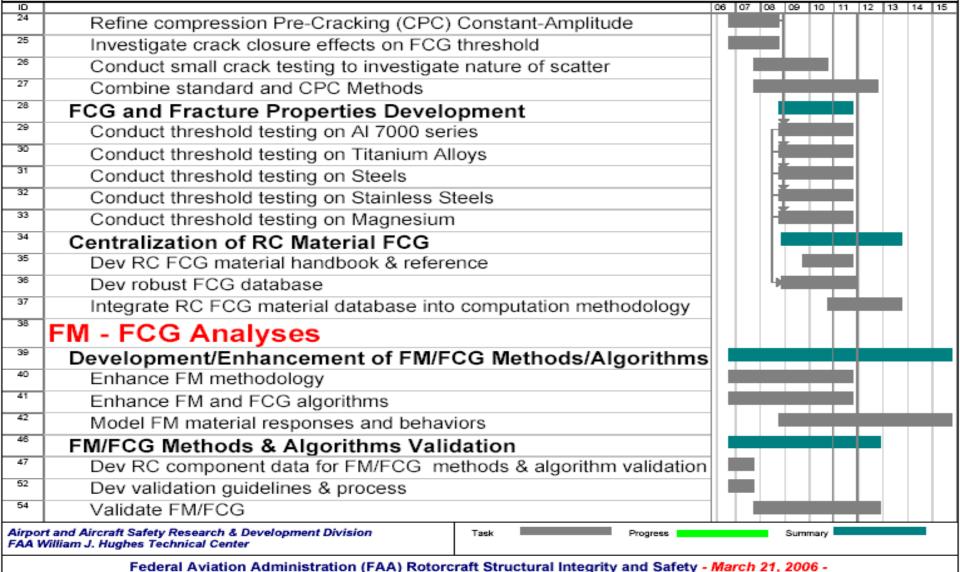
Federal Aviation Administration (FAA) Rotorcraft Structural Integrity and Safety - March 21, 2006 -

Task

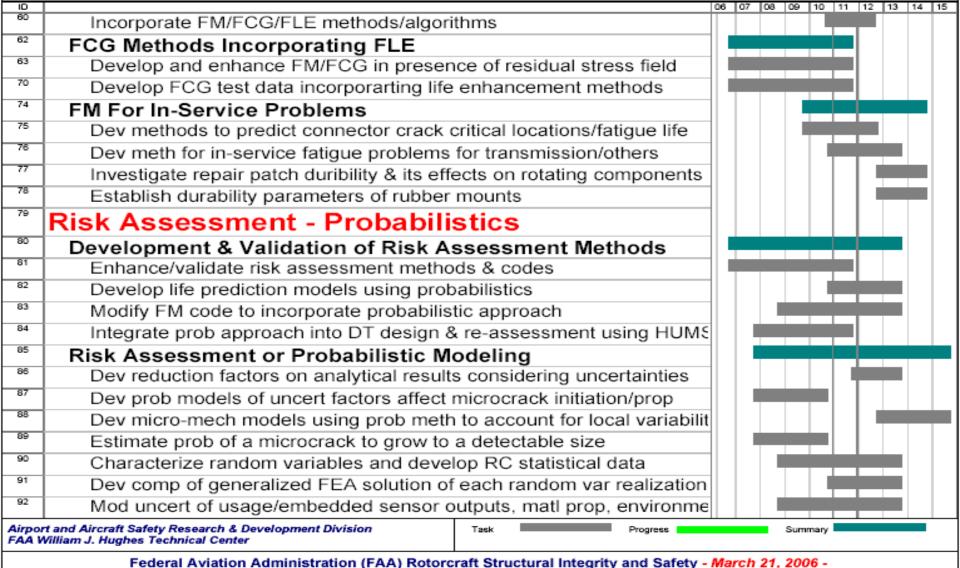


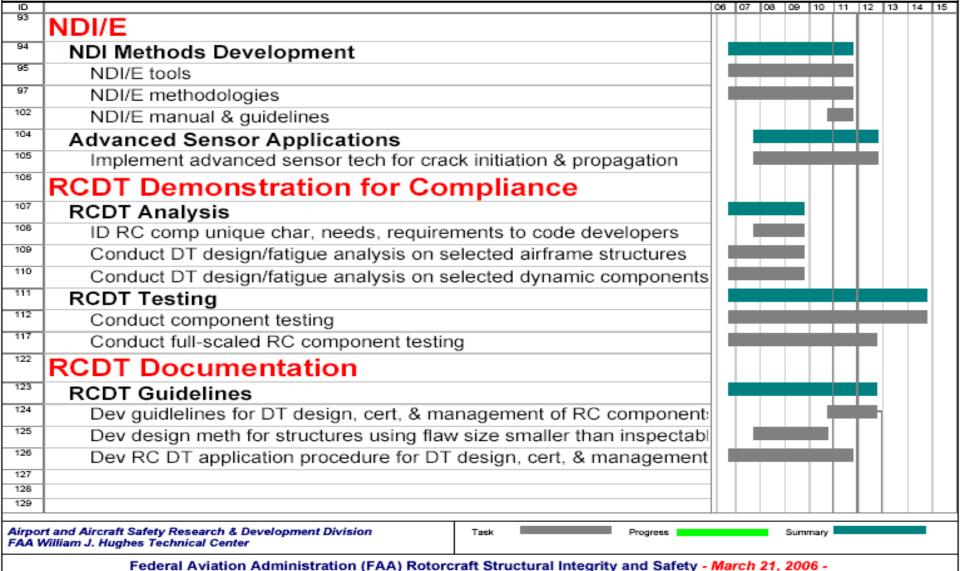
Airport and Aircraft Safety Research & Development Division

FAA William J. Hughes Technical Center



Federal Aviation Administration







ID		06	07	08	09	10	11	12	13	14	15
130	Regulatory Implementation & Tech Transfer										
131	FAA Order 8110.9 Revision										
132	Update usage spectrum and load spectra					1					
133	FAA Airworthiness Directives On RHL										
134	Address RHL lifing issues										
135	FAA Advisory Circular (AC) on RCDT	1									
136	Revise RCDT ACs										
137	RCDT Tech Transfer, Application, & Cert Compliance										
138	Transfer FM/FCG algorithms & codes										
139	Transfer NDI/E tools, accessories, methods	1									

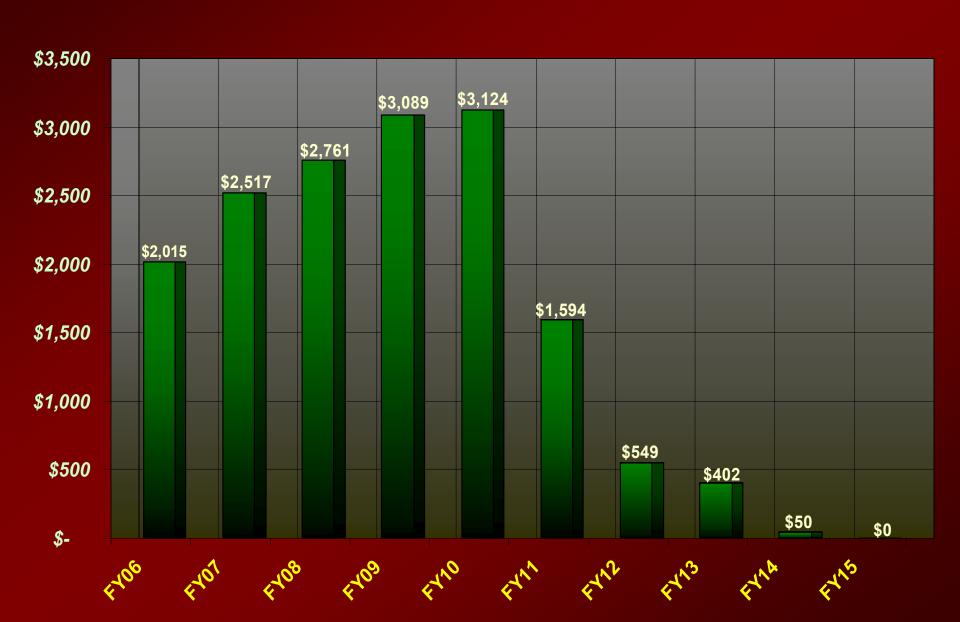
Airport and Aircraft Safety Research & Development Division FAA William J. Hughes Technical Center

Task

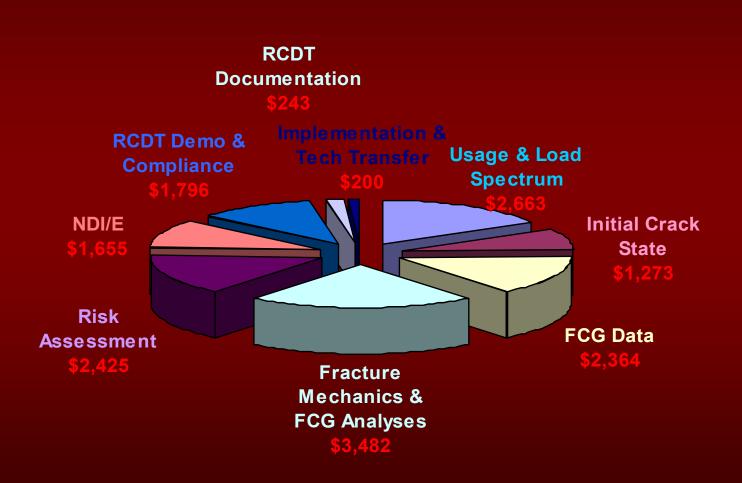
Progress Summary

Federal Aviation Administration (FAA) Rotorcraft Structural Integrity and Safety - March 21, 2006 -

FAA RCDT R&D 10-Year ROM (\$K) \$16M



FAA RCDT R&D 10-Year ROM (\$K) FY06 - FY15 \$16M







RCDT & HUMS BAA Status – As of 3/20/06

	RCDT	HUMS
Submitted WPs	21	17
Reviewed WPs	15	0
Requested for DTPs	3	0
Submitted DTPs	0	0
Reviewed DTPs	0	0
Accepted for Award	0	0